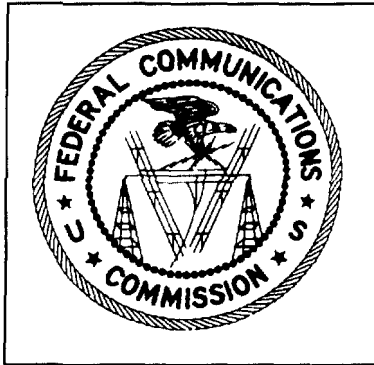


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**FEDERAL COMMUNICATIONS COMMISSION  
INTERNATIONAL BUREAU**

Satellite and Radiocommunication Division  
Satellite Policy Branch

**RECEIVED**

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**FEDERAL COMMUNICATIONS COMMISSION  
OFFICE OF SECRETARY**

**To:** Mr. William F. Caton, Acting Secretary

**Date:** February 6, 1996

**From:** Jennifer M. Gilsenan *JMG*

**Re:** Ex Parte presentation  
CC Docket No. 92-297

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This will serve to indicate that on February 5, 1996 representatives of the International Bureau, the Office of Plans and Policy and the Wireless Telecommunications Bureau met with participants listed in Attachment A to this memorandum to discuss the Commission's proposals in the above-referenced proceeding.

The attached documents formed a basis for discussion.

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List ABCDE

Attachment A

February 5, 1996

NAME	ORGANIZATION	PHONE	FAX
Harry Ng	FCC	202-418-0752	418-0761
Karl Kensing	FCC	202-418-0773	418-0765
THOMAS TYCZ	FCC	202-418-0735	418-0778
Don Gys	FCC	202-418-0734	202-418-0734
Bob James	FCC	202-418-0798	418-2643
David Wye	FCC	202-418-1897	418-0787
Rob Kubik	Motorola	602-732-6016	202-732-2305
David Carroll	Motorola	602-732-4918	602-732-2305
Jeffrey Kraus	Cellular Vision Vision Star	301-309-3703	301-309-9323
Charles Milkis	M. Gardner, PC - Cellular Vision	202/785-2828	-1504
Don Brittingham	Bell Atlantic	202-392-1189	202-392-1687
David G. O'Neill	Kastar	202/296-2007	202/429-0535
Richard Barnett	TS for Lockheed Martin	301-229-0204	301-320-2421
RAY BENDER	Dow, Johns + Albertson for Lockheed MARTIN	202-857-2758	202-857-2900
Philip Rubin	Pan AM SAT	202-296-9380	202-296-9383
Dan Goldberg	Goldberg, Godles, Wilner + Wright for PanAmSat	202-429-4900	202-429-4912
Jon Zeitler	Vinson & Elkins for NetSat 28	202-639-6562	202-639-6604
Tom Sullivan	NetSat 28	202-639-6562	202-639-6604
Steve Baruch	L, STL for TRW	202-416-6782	202-293-7783
Mike Horstein	TRW	310-812-2603	310-814-1400
Roger Rusch	TRW	310-812-2424	310-814-1400
NORM LEVENTHAL	L, STL for TRW	202-416-6744	202-293-7783
Andrew D'Uva	WILLIE FARR FOR CORAL	202-429-4773	202-887-8979
Barry Lemberman	Motorola	202-371-6929	202-842-3578
VICTOR BERNSTEIN	AT&T	908-221-5712	908-953-8360
Jim McCullough	AT&T	908-221-6957	908-221-4593

(over)

February 5, 1996

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# **BAND PLAN OPTIONS**

**February 5, 1996**



# **BAND PLAN PREFERENCES**



**1 GHz GSO/FSS spectrum is essential**

- **Supports mass market access**
- **Ubiquitous, low cost terminals**
- **Only 62% of international GSO FSS allocation**

**Objective is 1 GHz “clear” spectrum**

- **Already sharing with 2 degree adjacent GSOs**
    - **typical impact is 5-15% loss of capacity**
    - **e.g., Galaxy IV lost 7% of Ku capacity after extensive coordination**
  - **Willing to accept 1 GHz spectrum shared with NGSO MSS with certain constraints**
-

# **OPTION 3 NOT ACCEPTABLE**



**Inadequate bandwidth significantly hinders mass market business cases**

- 12 % reduction for GSO/FSS
- 17 % reduction for NGSO/MSS feeder links

**Penalizes efficient GSO/FSS multibeam architecture**

**Requires reworking of GSO/FSS system**

- Schedule delay
  - Increased cost
- Lost capacity
  - Smaller market
- Missed market opportunities

**Incompatible with WRC-95 and US position on GSO/NGSO sharing**

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# **GSO/NGSO SHARING IS POSSIBLE WITH CERTAIN CONSTRAINTS**



**Both Options 1 & 2 share GSO/FSS with NGSO/MSS  
feeder links**

**Hughes and TRW have worked to share spectrum for  
over a year**

**Sharing is feasible according to principles agreed  
upon by Hughes and TRW**

- Other GSO/FSS systems have similar capability

**Sharing imposes constraints on both GSO and NGSO**

- Requires separation of Iridium 150 MHz and shared GSO/NGSO
- Requires downlink band separation
- Constrains NGSO feeder link site selection
- Adversely affects GSO performance in several ways
  - e.g., link availability, capacity in shared beams





# Option 2 A

HUGHES  
COMMUNICATIONS

WRC '97

LMDS	GSO/FSS	NGSO/FSS	LMDS	GSO/FSS	GSO/FSS
fss	ngso/fss	gso/fss	MSS FL (Iridium &Odys)	MSS FL (Odyssey) (150 MHz)	ngso/fss
850 MHz	250 MHz	400 MHz	150 MHz	250 MHz	500 MHz
27.5	28.35	28.6 28.7	29.1	29.25	29.5
					30.0

- TRW/Odyssey and GSOs share 150 MHz without Iridium in same band
- TRW/Odyssey share 150MHz with Iridium
- Iridium and LMDS share 150 MHz
- Teledesic has conditional 400MHz without need to share and potential additional 100MHz allocated at WRC '97
- All get BW needed but LMDS must share with Iridium

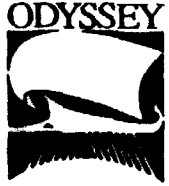
# Option 2 B

HUGHES  
COMMUNICATIONS

WRC '97

LMDS	GSO/FSS	L M D S	NGSO/FSS	MSS FL (Iridium &Odys)	GSO/FSS	GSO/FSS
fss	ngso/fss		gso/fss		MSS FL (Odyssey) (150 MHz)	ngso/fss
850 MHz	250 MHz	100 MHz	400 MHz	150 MHz	250 MHz	500 MHz
27.5	28.35	28.6 28.7		29.1 29.25	29.5	30.0

- LMDS return links moved to 28.6 to 28.7 GHz
- LMDS sharing with NGSO FSS (Iridium) avoided
- Teledesic BW constrained to 400 MHz - WRC '95 conditional allocation
- LMDS has 25 MHz less BW than option 3 – but clear
- All others stay the same – reduces sharing requirements on Iridium and Odyssey



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**CO-DIRECTIONAL FREQUENCY SHARING  
BETWEEN ODYSSEY™ FEEDER LINKS  
AND GSO/FSS SERVICE LINKS  
IN 29.25-29.5 GHZ AND 19.45-19.7 GHZ BANDS**

**M. Horstein  
R. Rusch**

**February 5, 1996**



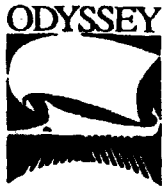
## Sharing Criteria



- **Acceptable interference due to NGSO/MSS feeder link at demodulator input in GSO/FSS system (CPM Report 95/118), based on interference/thermal noise spectral density ratio ( $I_o/N_o$ )**

Interference Level, $I_o$	Percent of Time Exceeded
Negligible ( $\leq 0.06$ )	0.87
0.78No	0.119
2.98No	0.0294
14.8No	0.0004

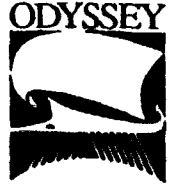
- **The same criteria will be assumed for FSS/GSO interference into NGSO/MSS feeder links**



## General Coordination Requirements



- The two interference mechanisms that can create a need for coordination are interference
  - From GSO/FSS earth terminals into Odyssey™ satellites
  - From Odyssey™ satellites into GSO earth terminals
- The need for coordination arises only for geometries in which an Odyssey™ earth station off-axis angle (i.e., the angle between lines of sight to an Odyssey™ satellite and a GSO/FSS satellite) is
  - Less than 1.0° for 0.66m GSO/FSS terminals, or
  - Less than 0.5° for 1.8m GSO/FSS terminals,and only if the elevation angle from the Odyssey earth station to the GSO/FSS satellite exceeds 30°
- When required, coordination applies only to GSO/FSS terminals located within a specified coordination area about the Odyssey™ earth station
- On a global basis, approximately 40% of the geostationary band requires no coordination with respect to any of the eight candidate Odyssey™ earth station locations

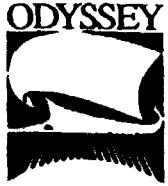


## **Coordination Requirements for GSO/FSS Satellites Providing U. S. Service**

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- **13 applied-for GSO slots from six different systems were examined to determine coordination requirements**
- **Based on an off-axis-angle criterion of  $1^\circ$  and an assumed satellite phasing, Odyssey<sup>TM</sup> candidate U. S. earth stations have the following coordination requirements:**
  - **3 of 13 GSO slots for San Luis Obispo, CA**
  - **5 of 13 GSO slots for Portland, ME**
- **There is generally one “encounter” daily between each GSO satellite/ Odyssey<sup>TM</sup> earth station pair requiring coordination**
- **For a required off-axis angle of  $1^\circ$ , each encounter lasts for 25 to 110 seconds, depending on the degree of “in-line” coincidence**



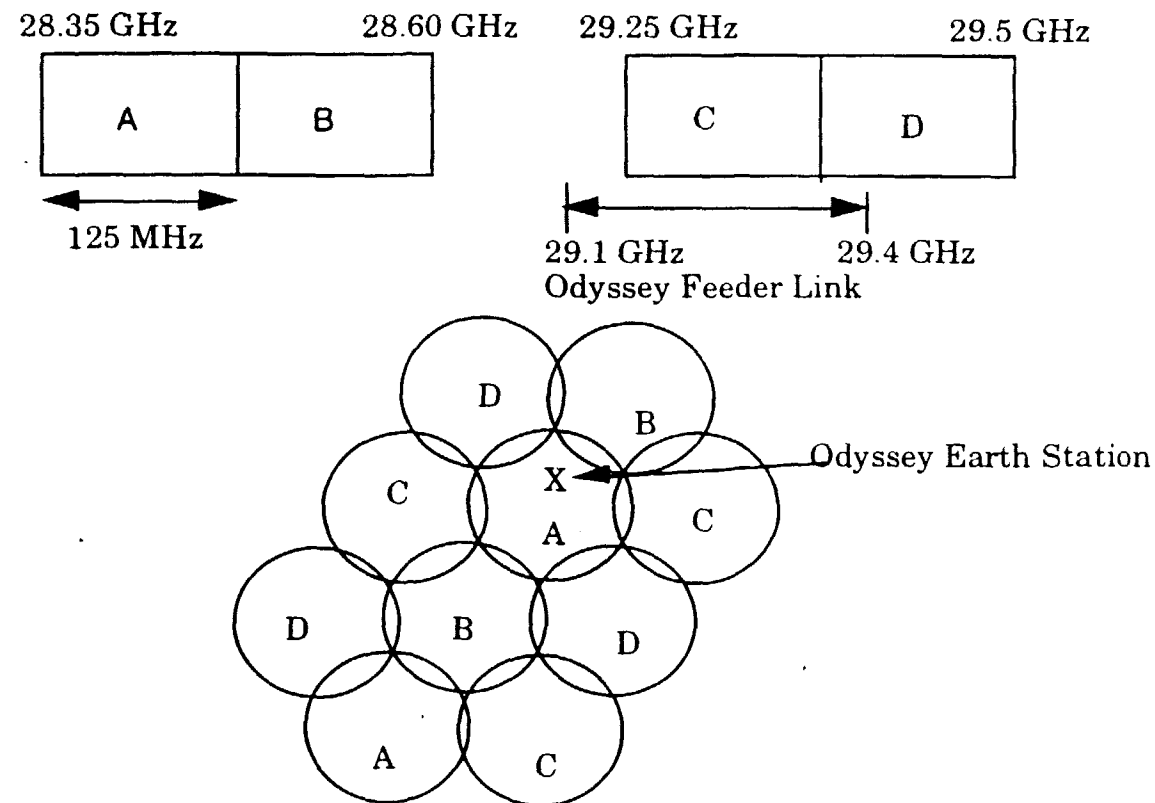
## Approach to Spectrum Sharing



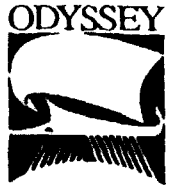
- **Spectrum sharing between Odyssey™ and GSO/FSS systems is predicated on**
  - **Single polarization of Odyssey™ transmissions**
  - **Frequency reuse plans (including both polarizations) of GSO/FSS systems**
- **Coordination can take the form of GSO/FSS system choosing disjoint frequency band or opposite polarization from Odyssey™ in beam(s) overlapping coordination zone**
- **Coordination can be aided by**
  - **Modification to Odyssey™ satellite phasing**
  - **Slight shift in GSO/FSS satellite antenna footprint**
- **It is anticipated that coordination as described above will preclude the need for Odyssey™ or GSO/FSS systems to interrupt transmissions**



# Co-Polarization Frequency Plan to Avoid Interference







## **Spectrum Sharing Principles Adopted by TRW and Hughes**



- **Sharing principles apply to the 29.25-29.4 GHz band and, following WRC-97, also to the 29.4-29.5 GHz band**
- **The party causing unacceptable interference has primary responsibility to mitigate the interference, but neither system shall be required to disrupt or alter its transmissions.**
- **Once Odyssey™ earth station locations and the frequency/polarity of its feeder link transmissions are defined, GSO/FSS operators will implement beam frequency selection and/or opposite sense of polarization in the vicinity of Odyssey™ earth station complexes in order to minimize instances of unacceptable interference in a manner that is consistent with their GSO beam footprint.**
- **Only NGSO/MSS systems may use the 19.3-19.7 GHz band as a companion downlink band to the 29.1-29.5 GHz uplink band. GSO/FSS systems may use any part of the 17.7-18.55 GHz band as a companion downlink band to the 29.25-29.5 GHz band.**
- **This sharing arrangement is only applicable to GSO/FSS operators and one type of NGSO/MSS system (namely, Odyssey™). An additional NGSO/MSS system will be accommodated only subject to further sharing and coordination agreements that are acceptable to the affected parties. Nevertheless, the sharing principles identified herein can be considered in attempting to reach such an accommodation.**

# Uplink Interference at Ka-Band from MSS Feeder Links (Iridium) into GSO FSS Systems

Dr. Richard Barnett

**TELECOMM STRATEGIES** for Lockheed Martin

February 5, 1996

# Assumptions (1)

- Worst case:
  - co-frequency (full frequency overlap)
  - co-polar (both operating in same polarization)
  - co-coverage (GSO receive beam peak at Iridium earth station)
- $I_0/N_0$  criteria:
  - CPM Report 95/118 (although not an ITU-R Rec.)
- Interferer as defined by Iridium:
  - transmit power density (peak) = -54.4 dBW/Hz
  - transmit earth station gain (peak) = +56.3 dBi
  - off-axis gain envelope < 29-26 log(theta) dBi

# Assumptions (2)

- GSO satellite receive characteristics:
  - peak antenna gain = 43.4 dBi (approximately  $1^\circ$  diameter)
  - system noise temperature = 600K (i.e., +27.8 dB-K)
  - gives  $G/T = +15.6$  dB/K
- Above beam is compatible with the use of 65 cm user terminals
- Receive antenna beam peak pointing towards Iridium uplink earth station location

# In-Line Interference Analysis

	IRIDIUM	ODYSSEY
Interfering power density into uplink antenna	-54.4 dBW/Hz	-55.49 dBW/Hz
Peak gain of uplink antenna	+56.3 dBi	+64.8 dBi
Peak interfering EIRP spectral density	+1.9 dBW/Hz	+9.31 dBW/Hz
Space loss	-213.5 dB	-213.5 dB
GSO satellite peak gain	+43.4 dB	+43.4 dB
GSO received interfering power density ( $I_0$ )	-168.2 dBW/Hz	-160.79 dBW/Hz
GSO satellite noise temperature	+27.8 dBK (600K)	+27.8 dBK (600K)
Boltzmann's constant	-228.6 dB	-228.6 dB
GSO receive noise power density ( $N_0$ )	-200.8 dBW/Hz	-200.8 dBW/Hz
GSO receive $I_0/N_0$	+32.6 dB	+40.01 dB

# Interference Threshold Off-Axis Angle

CPM criteria (95/118)			Computed isolation requirements for interference threshold			
			IRIDIUM		ODYSSEY	
% time not to be exceeded	$I_0/N_0$ (linear)	$I_0/N_0$ (dB)	Off-axis isolation	Off-axis angle $29-26\log(\theta)$	Off-axis isolation	Off-axis angle $29-25\log(\theta)$
0.87	0.06	-12.22	44.82 dB	4.72°	52.23 dB	4.54°
0.119	0.78	-1.08	33.68 dB	1.76°	41.09 dB	1.63°
0.0294	2.98	+4.74	27.86 dB	1.05°	35.27 dB	0.95°
0.0004	14.8	+11.70	20.90 dB	0.57°	28.31 dB	0.50°

# Iridium Simulation

- Assume feeder link earth station tracks all visible Iridium satellites (not just highest elevation satellite).
- Statistical results for each Iridium feeder link earth station site in the GSO/FSS coverage area may need to be summed to compute aggregate uplink interference.
- 30 day simulation takes <10 minutes computation time. (results stable after less than 10 days simulation).

# Iridium Constellation Definition

Spacecraft System Orbit Definition  
Ephemeris Data

7159.0	7159.0	7159.0	7159.0
0.00130	0.00130	0.00130	0.00130
86.400	86.400	86.400	86.400
345.631	345.631	345.631	345.631
331.772	3.362	34.952	66.472
181.376	197.736	214.096	230.456
11	11	11	11

90.0  
90.0

90.0  
90.0

Iridium

Spacecraft System Orbit Definition  
Ephemeris Data

90.0  
90.0

90.0  
90.0

Iridium



# Simulation Results ( $I_0/N_0 = 0.06$ )

